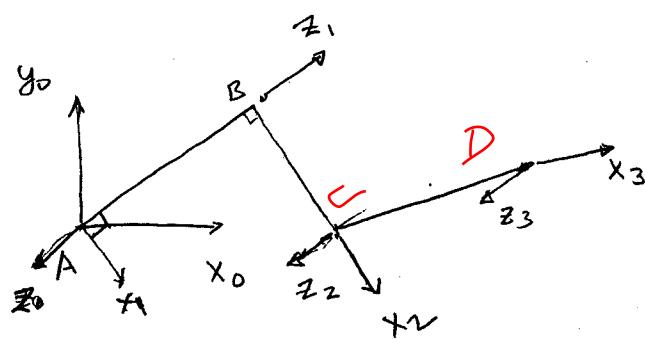
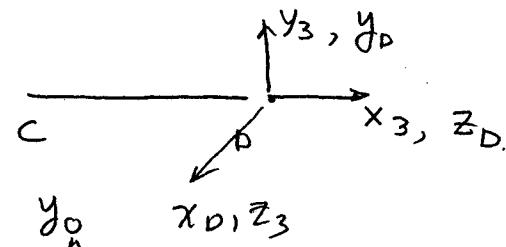


Q1 (a)

(i)



Last Link



$z_0, z_2 \}$ out of paper
 z_3

at $\theta_1 = -90^\circ$

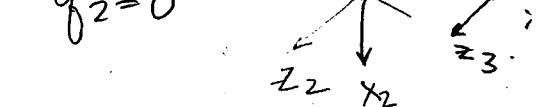
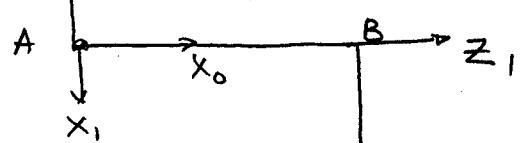
$$\theta_0 = \phi_1$$

(ii)

	θ	r	d	α
1	$\theta_1 = 0$	0	0	-90 ⁰
2	$\theta_2 = 1$	1	1	90 ⁰
3	$\theta_3 = 0$	0	1	0°

$$\theta_2 = 0^\circ$$

$$\theta_3 = +90^\circ$$



(iii) $\dot{\theta}_1 = \dot{\phi}_1 + (-90^\circ)$

$$\dot{\theta}_2 = \dot{\phi}_2$$

$$\dot{\theta}_3 = \dot{\phi}_3$$

(iv). Frame 0 + U.



$${}^0 T_0 = \begin{pmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}; {}^3 T_D = \begin{pmatrix} 0 & 0 & 1 & t \\ 0 & 1 & 0 & 0 \\ 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix},$$

$${}^U T_D = {}^U T_0 {}^0 T_1 {}^1 T_2 {}^2 T_3 {}^3 T_D.$$

$${}^U Z_D = {}^U T_D \begin{pmatrix} 0 \\ 0 \\ 0 \\ 1 \end{pmatrix}; {}^U P_D = {}^U T_D \begin{pmatrix} 0 \\ 0 \\ 1 \\ 1 \end{pmatrix}$$

Q1
(a)

$${}^A T_{B_0} = \text{initial position of } B = {}^A T_B = \text{given}$$



$${}^A T_{B_1} = \text{Rot}(z, 30^\circ) {}^A T_{B_0}$$

$${}^A T_{C_1} = {}^A T_{B_1} {}^{B_1} T_{C_1} \quad \text{where } {}^{B_1} T_{C_1} = {}^{B_1} T_C = \text{given}$$

$${}^A T_{C_2} = {}^A T_{C_1} \text{ Rot}(x, 40^\circ)$$

$${}^D T_{C_2} = {}^D T_A {}^A T_{C_2} \quad \text{where } {}^D T_A = {}^A T_D^{-1} = \text{given}$$

$${}^D T_{C_3} = \text{Rot}(y, 50^\circ) {}^D T_{C_2}$$

$${}^D T_{B_3} = {}^D T_{C_3} {}^{C_3} T_{B_3} \quad \text{where } {}^{C_3} T_{B_3} = \text{given} \\ = {}^B T_C$$

$${}^A T_{B_3} = {}^A T_D {}^D T_{B_3} \quad //$$

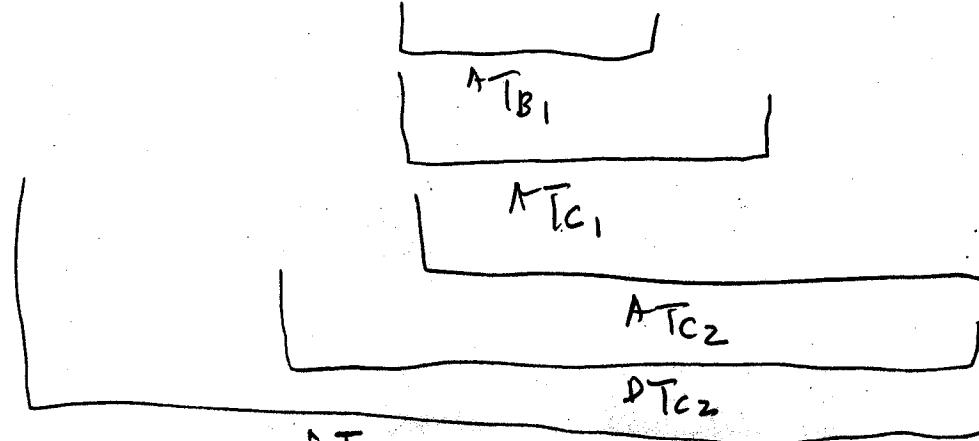
or

$${}^A T_{B_3} = {}^A T_D \text{ Rot}(y, 50^\circ) {}^D T_{C_2} {}^C T_B$$

$$= {}^A T_D \text{ Rot}(y, 50^\circ) {}^D T_A {}^A T_{C_2} {}^C T_B.$$

$$= {}^A T_D \text{ Rot}(y, 50^\circ) {}^D T_A {}^A T_{C_1} \text{ Rot}(x, 40^\circ) {}^C T_B$$

$$= {}^A T_D \text{ Rot}(y, 50^\circ) {}^D T_A \text{ Rot}(z, 30^\circ) {}^A T_{B_0} {}^B T_C \text{ Rot}(x, 40^\circ) {}^C T_B$$



Q2 (a)

For $\sin \beta \neq n_x \neq 0$

$$\theta_x = \sin \beta \sin \gamma, \quad a_x = \sin \beta \cos \gamma$$

$$\gamma = \text{ATAN2}(\theta_x, a_x) \quad \text{if } \sin \beta > 0$$

$$= \text{ATAN2}(-\theta_x, -a_x) \quad \text{if } \sin \beta < 0$$

$$\alpha = \text{ATAN2}(n_y, -n_z) \quad \text{if } \sin \beta > 0$$

$$= \text{ATAN2}(-n_y, n_z) \quad \text{if } \sin \beta < 0$$

$$\beta = \text{ATAN2}(\pm \sqrt{1 - n_x^2}, n_x)$$

$$\text{If } n_x = 1 \rightarrow \beta = 0^\circ$$

$$R = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(\alpha + \gamma) & -\sin(\alpha + \gamma) \\ 0 & \sin(\alpha + \gamma) & \cos(\alpha + \gamma) \end{pmatrix}$$

$$\alpha + \gamma = \text{ATAN2}(\theta_z, \theta_y) \quad \left\{ \text{infinitely many solns for } \alpha + \gamma \right.$$

$$\text{If } n_x = -1 \rightarrow \beta = 180^\circ$$

$$R = \begin{pmatrix} -1 & 0 & 0 \\ 0 & \cos(\alpha - \gamma) & \sin(\alpha - \gamma) \\ 0 & \sin(\alpha - \gamma) & -\cos(\alpha - \gamma) \end{pmatrix}$$

$$\alpha - \gamma = \text{ATAN2}(\theta_z, \theta_y) \quad \left\{ \text{infinitely many solns for } \alpha - \gamma \right.$$

2 solns

$$\beta = \text{ATAN2}(\pm \sqrt{1 - n_x^2}, n_x)$$

$$\alpha = \text{ATAN2}(n_y, -n_z)$$

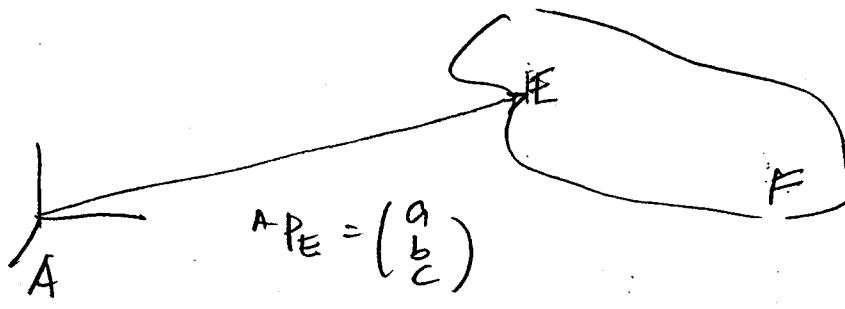
$$\gamma = \text{ATAN2}(\theta_x, a_x)$$

$$\beta = -\text{ATAN2}(-\sqrt{1 - n_x^2}, n_x)$$

$$\alpha = \text{ATAN2}(n_y, n_z)$$

$$\gamma = \text{ATAN2}(\theta_x, -a_x)$$

Q2 (b)



$${}^A P_E = \begin{pmatrix} a \\ b \\ c \end{pmatrix}$$

$${}^A R_E = \text{Rot}(x, \alpha) \text{Rot}(y, \beta) \text{Rot}(z, \gamma)$$

$${}^A w_F = {}^A w_E = \begin{pmatrix} 1 \\ 0 \\ 0 \end{pmatrix} \dot{\varphi} + \text{Rot}(x, \alpha) \begin{pmatrix} 0 \\ 1 \\ 0 \end{pmatrix} \dot{\beta} + \text{Rot}(x, \alpha) \text{Rot}(y, \beta) \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix} \dot{\gamma}$$

$${}^A u_F = {}^A u_E + {}^A w_F \times ({}^A R_E {}^E P_E)$$

//

$$\text{where } {}^E T_F = \begin{pmatrix} {}^E R_F & {}^E P_F \\ 0 & 0 \end{pmatrix}$$



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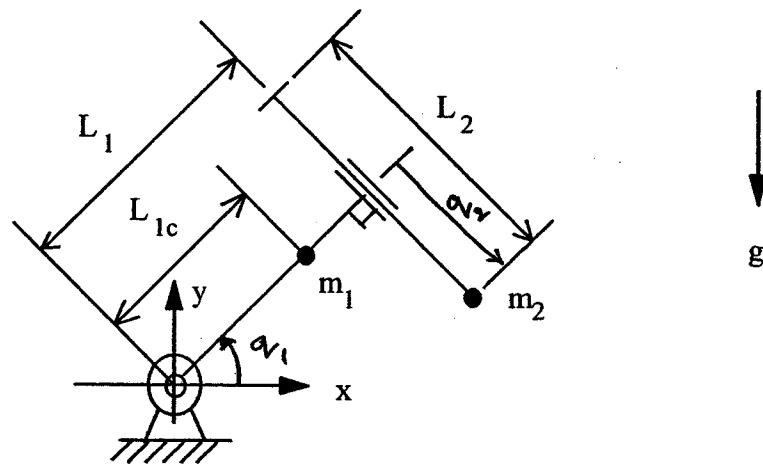
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Examination Solutions for:

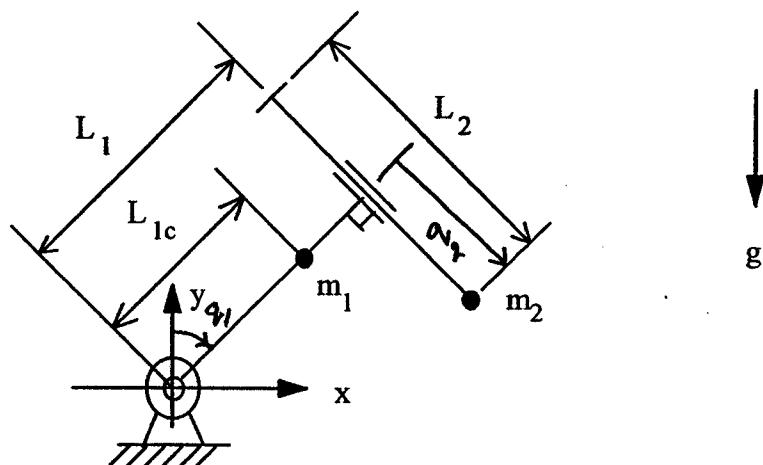
Course No. and Title:	EE4304/ME4245
Semester/Year:	Semester 1, 1999
Solution to Qn No. (& Marks):	3 (25 marks)

Q.3 (a)

Set 1.



Set 2.



The first joint is rotational and the second is prismatic.

Q. 3 (b) For set 1, the positions of m_1 and m_2 are:

$$m_1 \begin{cases} x_1 = l_{ic} \cos q_1 \\ y_1 = l_{ic} \sin q_1 \end{cases}$$

$$m_2 \begin{cases} x_2 = l_1 \cos q_1 + q_2 \sin q_1 \\ y_2 = l_1 \sin q_1 - q_2 \cos q_1 \end{cases}$$

The velocities of m_1 and m_2 are:

$$\dot{x}_1 = -l_{ic} \sin q_1 \dot{q}_1$$

$$\dot{y}_1 = l_{ic} \cos q_1 \dot{q}_1$$

$$\dot{x}_2 = -l_1 \sin q_1 \dot{q}_1 + q_2 \cos q_1 \dot{q}_2 + \dot{q}_2 \sin q_1$$

$$\dot{y}_2 = l_1 \cos q_1 \dot{q}_1 + q_2 \sin q_1 \dot{q}_2 - \dot{q}_2 \sin q_1$$

Giving:

$$v_1^2 = \dot{x}_1^2 + \dot{y}_1^2 = l_{ic}^2 \dot{q}_1^2$$

$$\begin{aligned} v_2^2 &= \dot{x}_2^2 + \dot{y}_2^2 = l_1^2 \dot{q}_1^2 + q_2^2 \dot{q}_1^2 + \dot{q}_2^2 - 2l_1 \dot{q}_1 \dot{q}_2 \\ &= l_1^2 \dot{q}_1^2 + \dot{q}_2^2 - 2l_1 \dot{q}_1 \dot{q}_2 + q_2^2 \dot{q}_1^2 \end{aligned}$$

$$= (l_1^2 + q_2^2) \dot{q}_1^2 + \dot{q}_2^2 - 2l_1 \dot{q}_1 \dot{q}_2$$

$$= (l_1 \dot{q}_1 - \dot{q}_2)^2 + q_2^2 \dot{q}_1^2$$

Q.3 (b) Continued:

The total kinetic energy of the system is:

$$\begin{aligned} K &= \frac{1}{2} m_1 \omega_1^2 + \frac{1}{2} m_2 \omega_2^2 \\ &= \frac{1}{2} m_1 l_{ic}^2 \dot{\varphi}_1^2 + \frac{1}{2} m_2 [(l_1^2 + q_2^2) \dot{\varphi}_1^2 + \dot{q}_2^2 - 2 l_1 \dot{\varphi}_1 \dot{q}_2] \\ &= \frac{1}{2} \left\{ [m_1 l_{ic}^2 + m_2 (l_1^2 + q_2^2)] \dot{\varphi}_1^2 - 2 l_1 \dot{\varphi}_1 \dot{q}_2 + m_2 \dot{q}_2^2 \right\} \\ &= \frac{1}{2} \dot{q}^T D \dot{q} \end{aligned}$$

where,

$$D(q) = \begin{bmatrix} d_{11} & d_{12} \\ d_{21} & d_{22} \end{bmatrix},$$

with. $d_{11} = m_1 l_{ic}^2 + m_2 (l_1^2 + q_2^2)$

$$d_{12} = d_{21} = -m_2 l_1,$$

$$d_{22} = m_2$$

The potential energy of the system is:

$$P = l_{ic} m_1 g \sin \varphi_1 + m_2 g (l_1 \sin \varphi_1 - q_2 \cos \varphi_1)$$

The Christoffel symbols defined by,

$$\gamma_{ijk} = \frac{1}{2} \left[\frac{\partial d_{kj}}{\partial q_i} + \frac{\partial d_{ki}}{\partial q_j} - \frac{\partial d_{ij}}{\partial q_k} \right],$$

Q. 3 (b) (Continued):

are given by:

$$C_{111} = \frac{1}{2} \frac{\partial d_{11}}{\partial q_1} = 0$$

$$C_{121} = C_{211} = \frac{1}{2} \frac{\partial d_{11}}{\partial q_2} = m_2 \dot{q}_2$$

$$C_{221} = \frac{\partial d_{12}}{\partial q_2} - \frac{1}{2} \frac{\partial d_{22}}{\partial q_1} = 0$$

$$C_{112} = \frac{\partial d_{21}}{\partial q_1} - \frac{1}{2} \frac{\partial d_{11}}{\partial q_2} = -m_2 \dot{q}_2$$

$$C_{122} = C_{212} = \frac{1}{2} \frac{\partial d_{22}}{\partial q_1} = 0$$

$$C_{222} = \frac{1}{2} \frac{\partial d_{22}}{\partial q_2} = 0$$

By definition: $C_{kj} = \sum_{i=1}^n C_{ijk} \dot{q}_i$,

giving:

$$C(q, \dot{q}) = \begin{bmatrix} m_2 \dot{q}_2 \dot{q}_2 & m_2 \dot{q}_2 \dot{q}_1 \\ -m_2 \dot{q}_1 \dot{q}_2 & 0 \end{bmatrix}$$

The gravitational force,

$$G(q) = \frac{\partial P}{\partial q} = \begin{bmatrix} m_1 g l_1 \cos q_1 + m_2 g (l_1 \cos q_1 + q_2 \sin q_1) \\ -m_2 g \cos q_1 \end{bmatrix}$$

The Complete dynamics is given by,

$$D(q) \ddot{q} + C(q, \dot{q}) \dot{q} + G(q) = \tau = [\tau_1 \ \tau_2]^T$$

Q. 3 (c)

Computed torque method:

$$\tau = D(q)\dot{v} + C(q, \dot{q})\dot{q} + G(q) \quad (1)$$

where,

 v is the new control signal to be defined.

Substituting the above control law into the robot dynamics, a double integrator system is obtained:

$$\ddot{q} = v \quad (2)$$

Define

$$v_i = -k_{vi}\dot{q}_i - k_{pi}q_i \quad (3)$$

This is a PD control scheme.

Substituting into the double integrator above, the closed loop system is obtained as:

$$\ddot{q}_i + k_{vi}\dot{q}_i + k_{pi}q_i = 0 ; \quad i=1,2$$

Taking Laplace transform:

$$s^2 + k_{vi}s + k_{pi} = 0$$

Comparing with the standard equation:

$$s^2 + 2\zeta\omega_n s + \omega_n^2 = 0$$

 \Rightarrow

$$k_{vi} = 2\zeta\omega_n$$

$$k_{pi} = \omega_n^2 = 4^2 = 16$$

For a critically damped system;
 $\zeta = 1$.

$$\therefore k_{vi} = 2 * 1 * 4 = 8.$$



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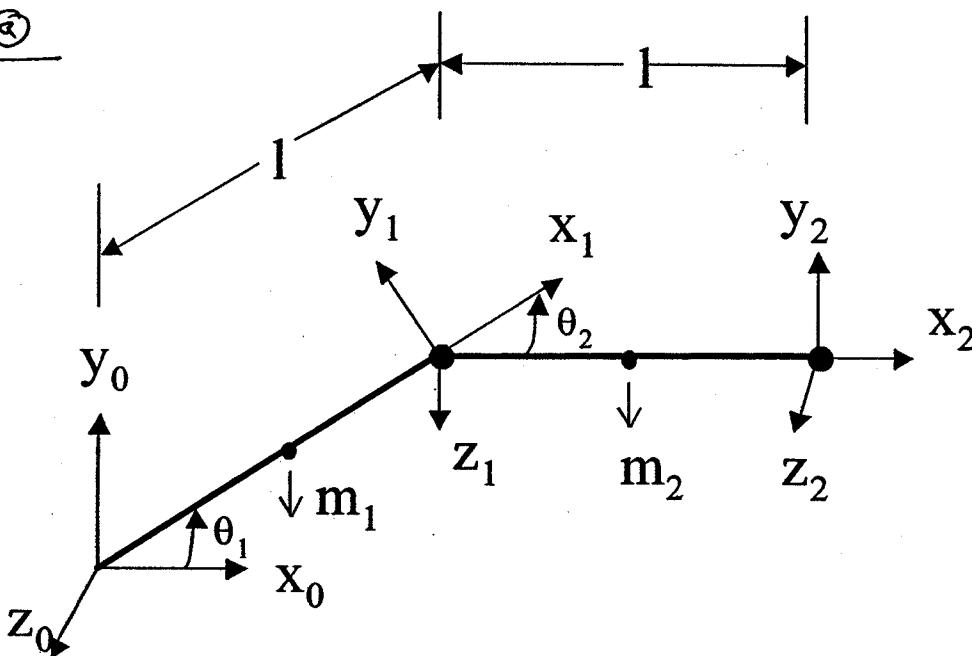
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DEPARTMENT OF ELECTRICAL ENGINEERING

Examination Solutions for:

Course No. and Title:	EE4304/ME4245
Semester/Year:	Semester 1, 1999
Solution to Qn No. (& Marks):	4 (25 marks)

Q 4.(a)



(i) Using the notation of the L-E equation of motion, we have :

$$D(\theta)\ddot{\theta}(t) + c(\theta, \dot{\theta}) + g(\theta) = \tau(t)$$

where

$$D(\theta) = \begin{bmatrix} d_{11}(\theta_2) & d_{12}(\theta_2) \\ d_{12}(\theta_2) & d_{22} \end{bmatrix},$$

$$c(\theta, \dot{\theta}) = \begin{bmatrix} c_{12}(\theta_2)(\dot{\theta}_2)^2 + 2c_{12}(\theta_2)\dot{\theta}_1\dot{\theta}_2 \\ -c_{12}(\theta_2)\dot{\theta}_1^2 \end{bmatrix} = \begin{bmatrix} c_1 \\ c_2 \end{bmatrix}$$

$$\text{and } g(\theta) = \begin{bmatrix} g_1(\theta_1, \theta_2) \\ g_2(\theta_1, \theta_2) \end{bmatrix}$$

Q. 4.(a)(i) Continued.

Since the inverse of $D(\theta)$ always exists (for any practical robot):

$$D^{-1}(\theta) = \frac{1}{d_{11}d_{22} - d_{12}^2} \begin{bmatrix} d_{22} & -d_{12}(\theta_2) \\ -d_{12}(\theta_2) & d_{11}(\theta_2) \end{bmatrix};$$

We have,

$$\begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \end{bmatrix} = -D^{-1}(\theta) \left[(C(\theta, \dot{\theta}) + g(\theta)) \right] + D^{-1}(\theta) \tau(t)$$

Choose the state variable vector as

$$x(t) = (\theta_1, \theta_2, \dot{\theta}_1, \dot{\theta}_2)^T \triangleq (x_1, x_2, x_3, x_4)^T$$

and the control vector as:

$$u(t) = (\tau_1, \tau_2)^T \triangleq (u_1, u_2)^T$$

Q. 4.(a)(ii) From the above equations, the state equations can be obtained:

$$\dot{x}_1(t) = \dot{\theta}_1 = x_3$$

$$\dot{x}_2(t) = \dot{\theta}_2 = x_4$$

$$\dot{x}_3(t) = \ddot{\theta}_1 = \frac{-1}{d_{11}d_{22} - d_{12}^2} \left[d_{22}(c_1 + g_1) - d_{12}(c_2 + g_2) - u_1 \right]$$

$$\dot{x}_4(t) = \ddot{\theta}_2 = \frac{-1}{d_{11}d_{22} - d_{12}^2} \left[-d_{12}(c_1 + g_1) + d_{11}(c_2 + g_2) - u_2 \right]$$

Q.4.(a)(ii) (Continued)

Expressed in matrix form, we have,

$$\frac{d}{dt} \begin{bmatrix} x_1(t) \\ x_2(t) \\ x_3(t) \\ x_4(t) \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \\ x_3(t) \\ x_4(t) \end{bmatrix}$$

$$- \frac{1}{d_{11}d_{22} - d_{12}^2} \begin{bmatrix} 0 \\ 0 \\ d_{22}(c_{12}\dot{\theta}_2^2 + 2c_{12}\dot{\theta}_1\dot{\theta}_2 + g_1) - d_{12}(-c_{12}\dot{\theta}_1^2 + g_2) \\ -d_{12}(c_{12}\dot{\theta}_2^2 + 2c_{12}\dot{\theta}_1\dot{\theta}_2 + g_1) + d_{11}(-c_{12}\dot{\theta}_1^2 + g_2) \end{bmatrix}$$

$$+ \frac{1}{d_{11}d_{22} - d_{12}^2} \begin{bmatrix} 0 \\ 0 \\ u_1(t) \\ u_2(t) \end{bmatrix}$$

Q.4.(b) The basic idea of independent joint control schemes is:

Assume that each joint is modeled by

$$\bar{m}_i \ddot{q}_i = \tau_i$$

where \bar{m}_i is the nominal inertia value of joint i .

Then the well known PID controller of the form,

$$\tau_i = k_{oi} \dot{q}_i + k_{pi} q_i + k_{ii} \int_0^t q_i dt$$

can be used.

Q. 4 (b) Continued

The advantages of the control scheme are:

- (i) Simple and easy to implement
- (ii) robust in terms of parameter variations.

The disadvantages of the ^{scheme} are:

- (i) The closed loop performance is not very good, as the coupling is not compensated for
- (ii) The control performance varies for different loads and different robot configurations.

Q. 4. (c)

Advantages: Conceptually simple, easy to design and, it is easy to specify the design performance

Disadvantages: Exact knowledge about the system is required, computationally intensive, and not robust to unmodeled dynamics.



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Examination Solutions for:

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Solution to Qn No. (& Marks):	5.(25 marks)

Q. 5 @ (i) 5th degree polynomial unless the trajectory is split into several trajectory segments. In this case, 3rd degree polynomial is the lowest.

(ii) For a quartic polynomial, we have 5 unknowns, but with 6 constraints. So it is required to split the trajectory into two segments. An intermediate point can be picked and with the constraints the unknowns can be found.

$$\begin{array}{ccccccc} & t_1 & & t_2 & & & \\ \hline X & & & & & & X \\ \theta_0 = 30^\circ & h_1(t) & & h_2(t) & & \theta_f = 100^\circ \\ v_0 = 0 & & & & & v_f = 0 \\ R_0 = 0 & & & & & R_f = 0 \\ & \dot{\theta}_1^- = \dot{\theta}_1^+ & & & & & \\ & R_1^- = R_1^+ & & & & & \end{array}$$

For each trajectory segments (2 segments) we have a quartic polynomial.

Q. 5 (a) ii: Continued

For the first segment:

$$h_1(t) = a_{14}t^4 + a_{13}t^3 + a_{12}t^2 + a_{11}t + a_{10}, \quad t \in [0, 1].$$

Using the position, velocity and acceleration constraints at $t=0$:

$$h_1(t) = a_{14}t^4 + a_{13}t^3 + a_0, \quad \text{where } a_0 = 30^\circ.$$

For the second segment:

$$h_n(t) = a_{n4}t^4 + a_{n3}t^3 + a_{n2}t^2 + a_{n1}t + a_{n0}, \quad t \in [-1, 0]$$

Using the position, velocity and acceleration constraints at $t=0$:

$$h_n(t) = a_{n4}t^4 + a_{n3}t^3 + a_f, \quad \text{where } a_f = 100^\circ.$$

To have continuity between segments, the velocity and acceleration from the first segment must be continuous with that at the beginning of the second segment:-

$$\dot{a}_1^- = \frac{\dot{h}_1}{t_1} \Big|_{t=1} = \frac{4a_{14} + 3a_{13}}{t_1}$$

$$\dot{v}_1^+ = \frac{\dot{h}_n}{t_n} \Big|_{t=-1} = \frac{-4a_{n4} + 3a_{n3}}{t_2}$$

Q. 5 (a)(ii). Continued

and,

$$q_i^- = \frac{h_i}{t_i^2} \Big|_{t=1} = \frac{12 q_{14} + 6 q_{13}}{t_1^2}$$

$$q_i^+ = \frac{h_i}{t_i^2} \Big|_{t=-1} = \frac{12 q_{n4} - 6 q_{n3}}{t_2^2}$$

Equating: $v_i^- = v_i^+$ and $q_i^- = q_i^+$, give two equations with 4 unknowns:

At the selected intermediate position, the difference of joint angles between the two segments are,

$$\delta_1 = h_i(1) - h_i(0) = q_{14} + q_{13}$$

$$\delta_n = h_n(0) - h_n(-1) = -q_{n4} + q_{n3}$$

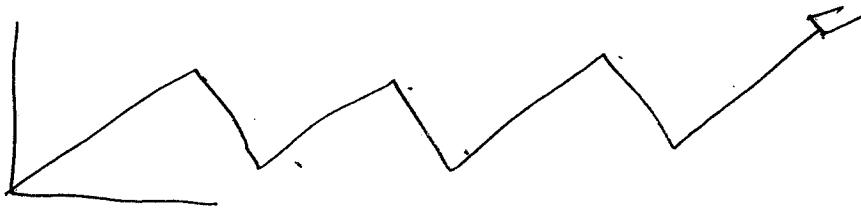
The 4 unknown equations can be expressed as:

$$\begin{bmatrix} 3/t_1 & 4/t_1 & -3/t_2 & 4/t_2 \\ 6/t_1^2 & 12/t_1^2 & 6/t_2^2 & -12/t_2^2 \\ 1 & 1 & 0 & 0 \\ 0 & 0 & 1 & -1 \end{bmatrix} \begin{bmatrix} q_{13} \\ q_{14} \\ q_{n3} \\ q_{n4} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ \delta_1 \\ \delta_n \end{bmatrix}$$

\Rightarrow give the solution.

For the ^{second} segment, the normalized time is from $t \in [-1, 0]$ and has to be changed back to $t \in [0, 1]$ by substituting $t = (t-1)$ in the expression for $h_n(t)$.

Q5(b)



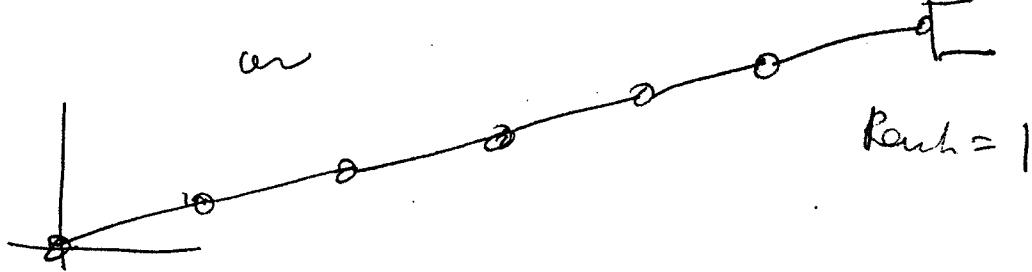
- (i) 3 dof
- 2 position (x, y)
 - 1 orientation about z

(ii) Rank < 3 at singular

(iii)



or



(iv) 3 joints